

Numerical Simulation of the Downstream Fan Noise of a Coaxial Jet with a Shielding Surface

*SEBU project (SEmi-BURied engine), co-funded by Airbus SAS and
EREA*

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*“Aeroacoustics of New Aircraft & Engine Configurations”, 11th -12th November 2004, Budapest
(Hungary)*

Objective

Comparative study of both a classical and a modified co-axial engine, in order to evaluate both the :

- installation shielding effect (secondary exhaust extension)
- highly inhomogeneous mean flow effect

on

the downstream rear tone fan noise

Isolated nozzle

Shielded nozzle

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General methodology

Preliminary RANS (3D/ 2D-axi) mean flow computation : CFD solver

V_0
→

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**Near & mid-field propagation: n:
Euler's solver**

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V_0
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Kirchhoff frontier

Kirchhoff frontier

Far field radiation : Kirchhoff's solver

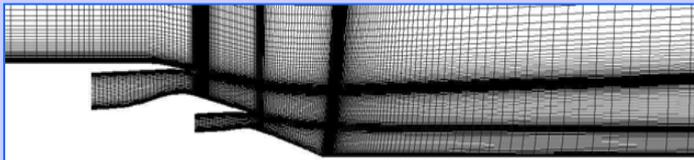
Near & mid-field propagation: n: Euler's solver

Outline

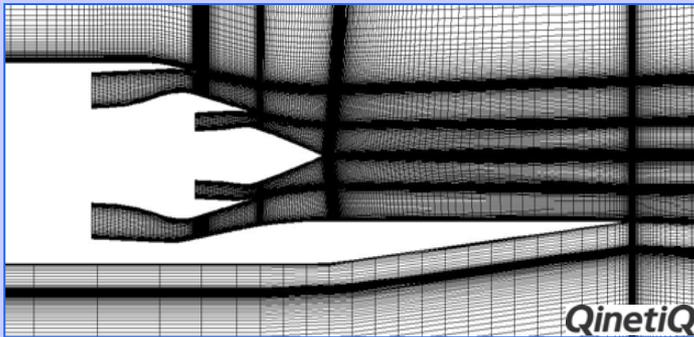
- ❑ **Preliminary tasks**
- ❑ **CAA Hybrid Methodology Tools**
- ❑ **2D study of the Shielding Effect in a Quiescent Medium**
- ❑ **2D study of the Shielding Effect in an Inhomogeneous Medium**
- ❑ **Conclusions of the 2D studies : Shielding and Mean Flow Effects**
- ❑ **Early 3D results : Isolated Nozzle in both a Quiescent and an Inhomogeneous Medium**
- ❑ **Conclusions & Perspectives**

Preliminary tasks

Preliminary CFD (RANS) Computations (*QinetiQ*, *Fluent v.5*)

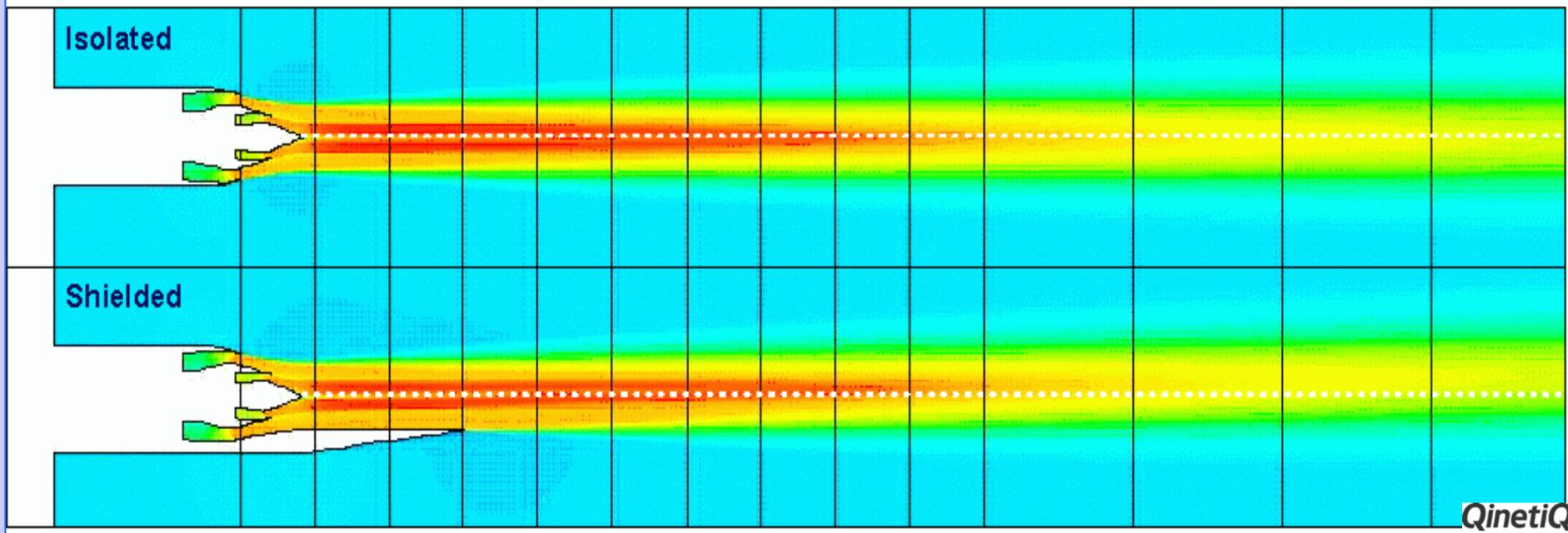


*Isolated axi-symmetric nozzle (2D mesh, partial view):
5 structured domains (94 805 elements)*



*Shielded nozzle (3D mesh, partial view):
10 structured domains (5 542 100 elements)*

High Subsonic Hot Jet, Take Off Conditions ($M^\infty = 0.25$ / $M^{max} \approx 0.9$) : Axial velocity results



CAA Hybrid Methodology Tools

Near & Mid-Fields Propagation over Inhomogeneous Flows & Solid Bodies : *sAbrinA*

("Solver for Acoustic BRoadband Interaction with Aerodynamics")

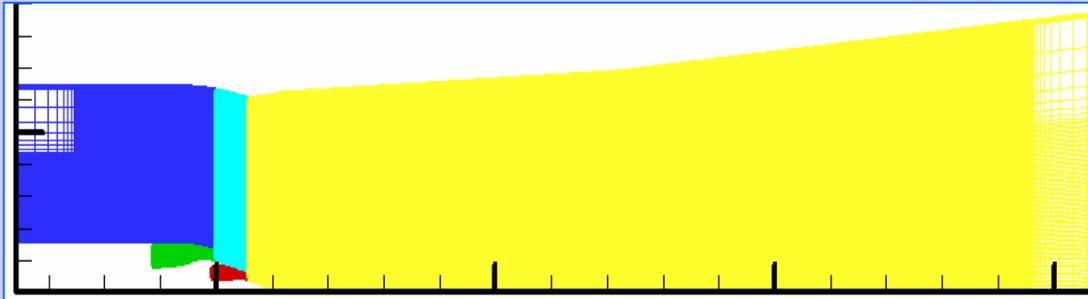
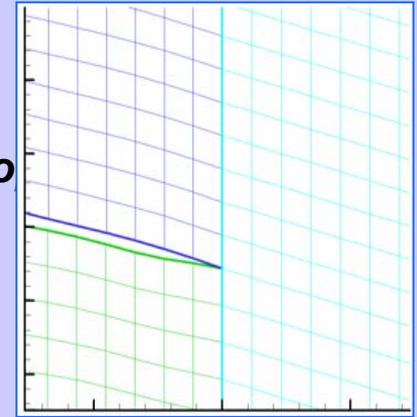
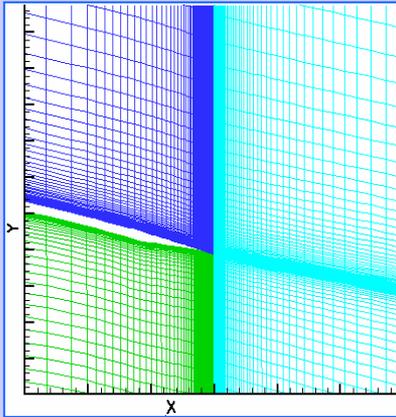
- * Full Euler or Navier-Stokes equations
- * Complete (NLDE) or Splitted variables (hybrid CAA)
- * Finite Differences or Volumes
- * High-order (up to 6th order in space, up to 3rd order in time, up to 10th order for filters)
- * Multi-D, multi-domain
- * Specific boundary conditions (rigid obstacle, symmetry plane, free-field)

Far Field Radiation in Homogeneous Medium : *Kirch2D, Kirch3D*

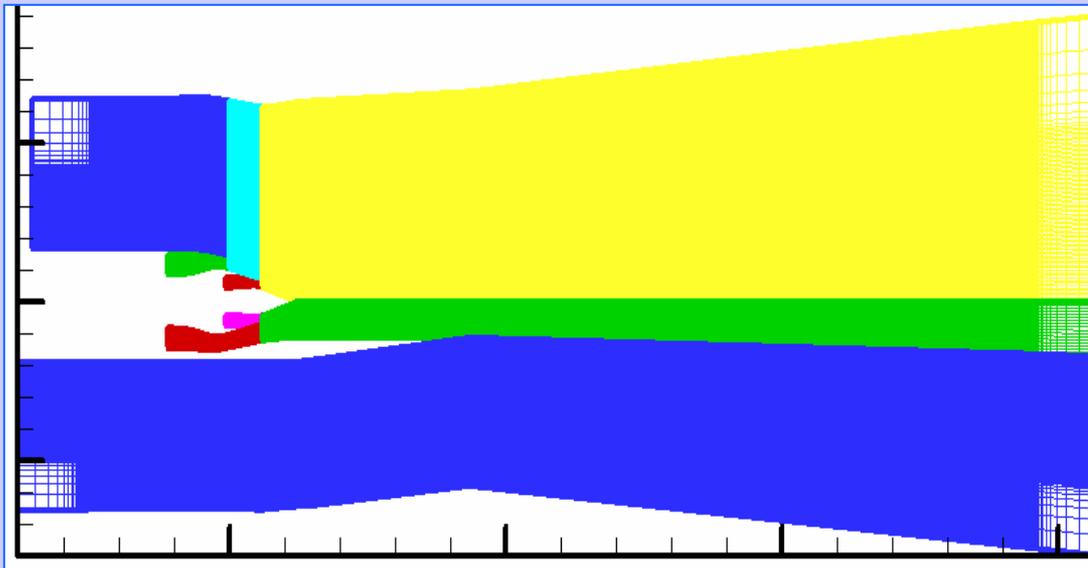
- * Frequential 2D or 3D codes (Kirchhoff integration)
- * Frequency or temporal, harmonic or broadband signals
- * Homogeneous medium (at rest or with uniform flow)

2D Acoustic Grid Derivation

- . Conservation of blocks number, frontiers and topology
- . Complete re-meshing of each block



*Isolated nozzle (2D mesh):
5 structured domains (64 408 elements)*

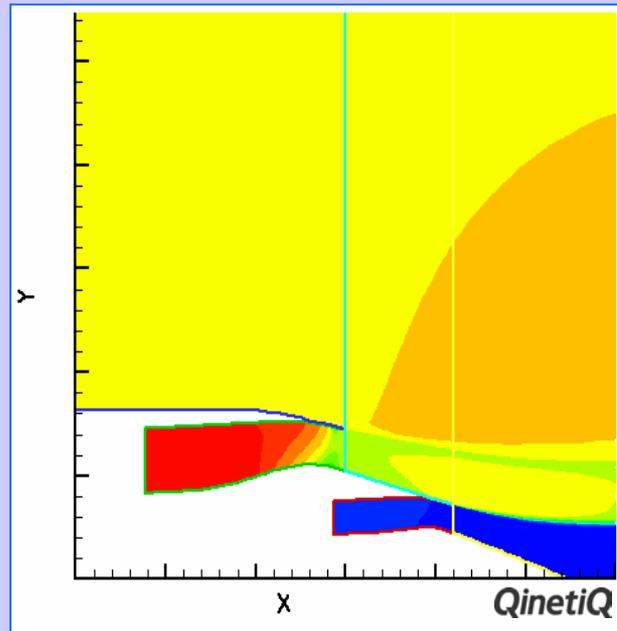


*Shielded nozzle (2D mesh):
9 structured domains (154 442 elements)*

$$\Delta y_{min\ CAA} = 500 \Delta y_{min\ CFD} = \lambda/10$$

2D RANS Mean Flow Interpolation (*Barycentral bilinear technique*)

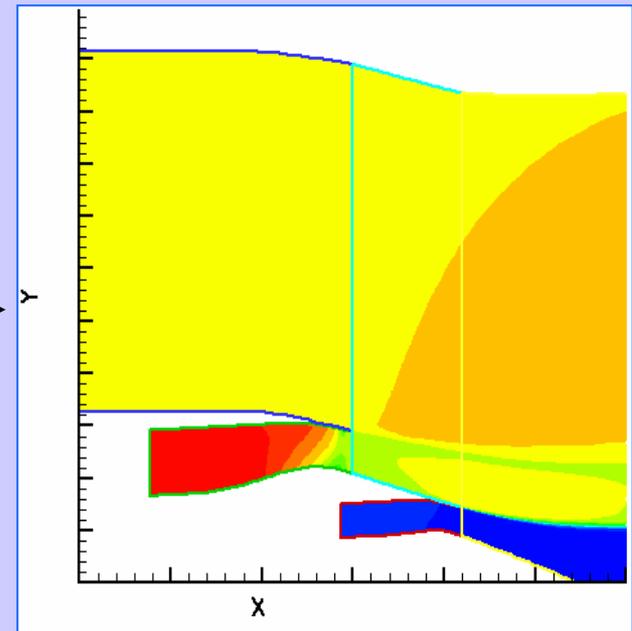
Density field on the CFD grid



*Barycentral
bilinear
technique*



Density field on the CAA grid



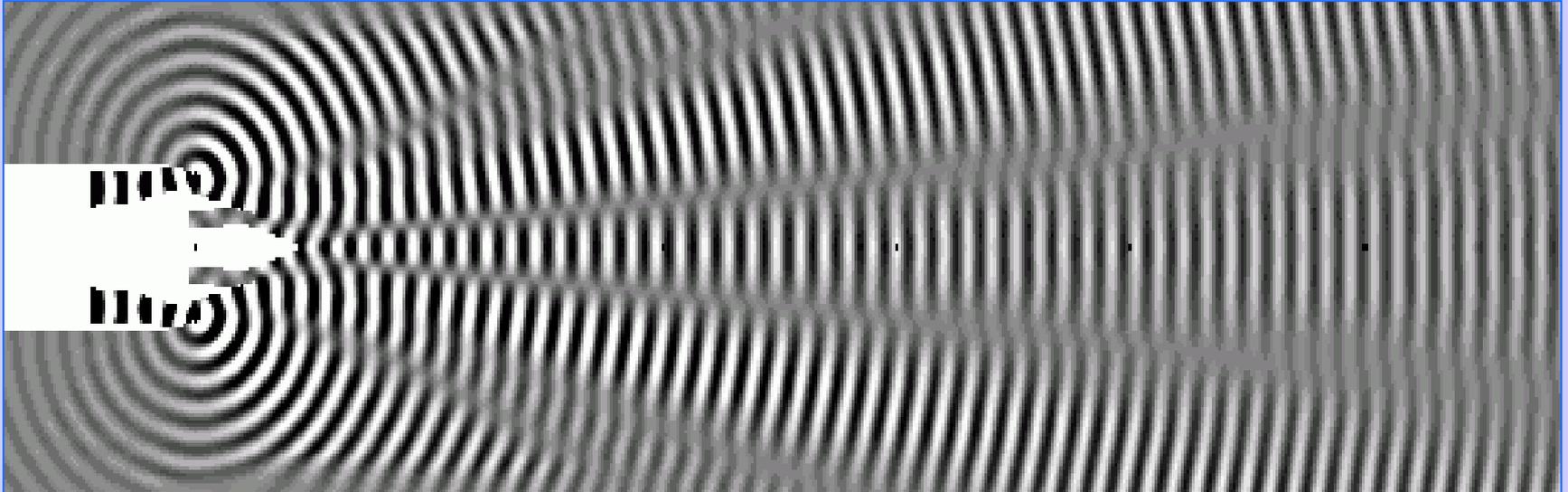
Shielding Effect on the Fan Noise 2D Propagation in a Quiescent Medium

Fan Noise in a Quiescent Medium

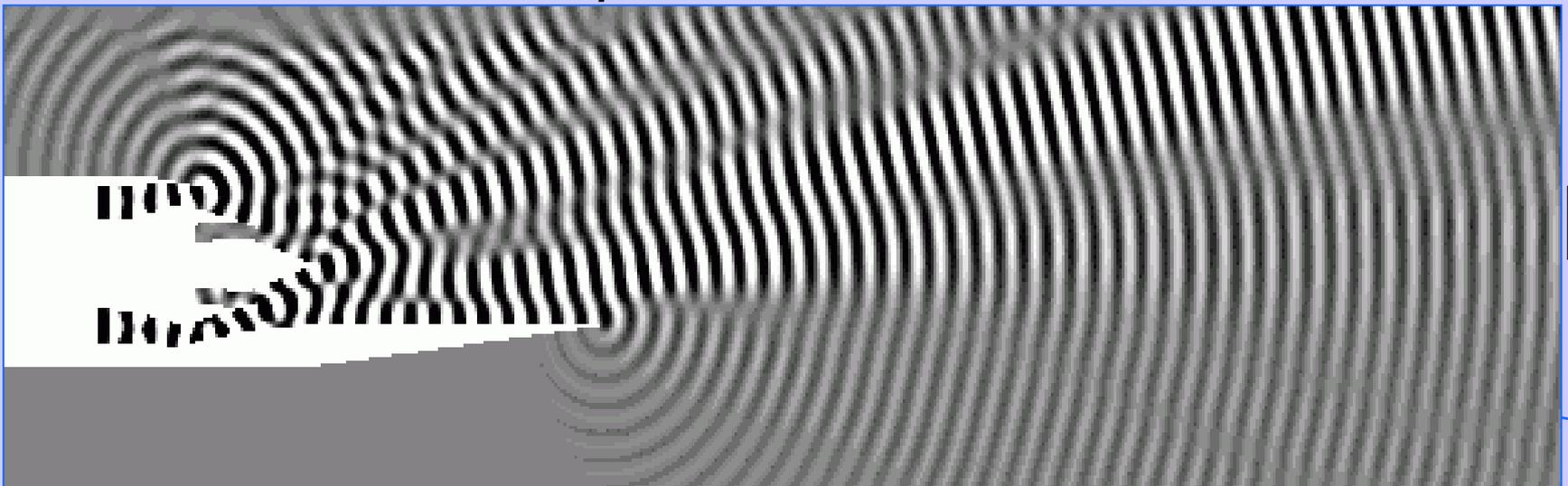
(1/3) Near & Mid-Fields 2D Propagation (sAbrinA code)

Isolated nozzle : Instantaneous pressure field

$kR = 20$



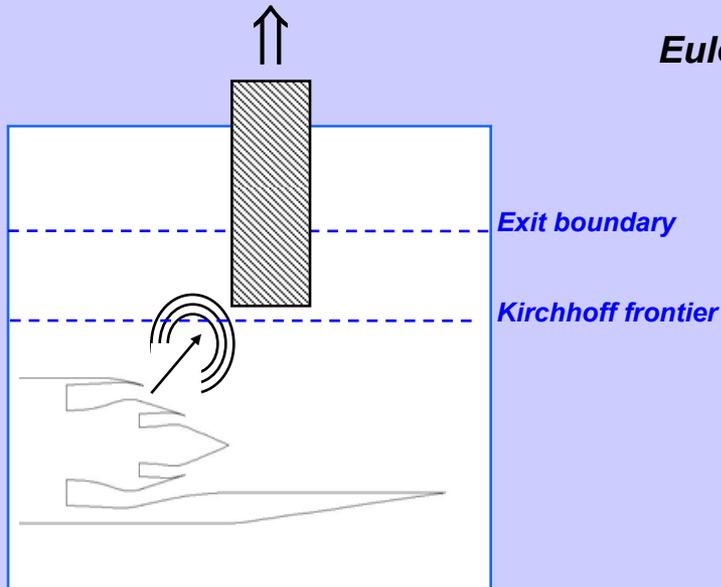
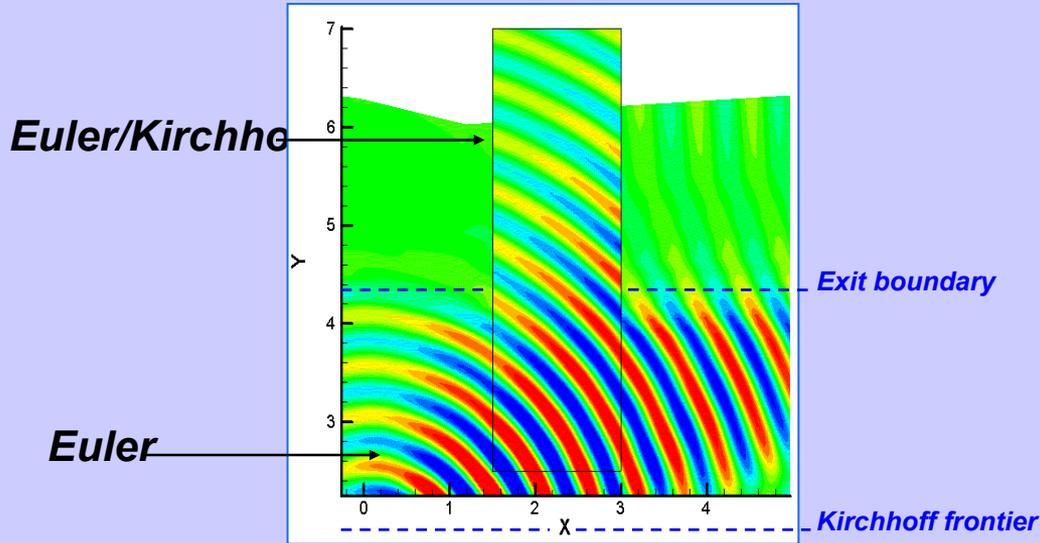
Shielded nozzle: Instantaneous pressure field



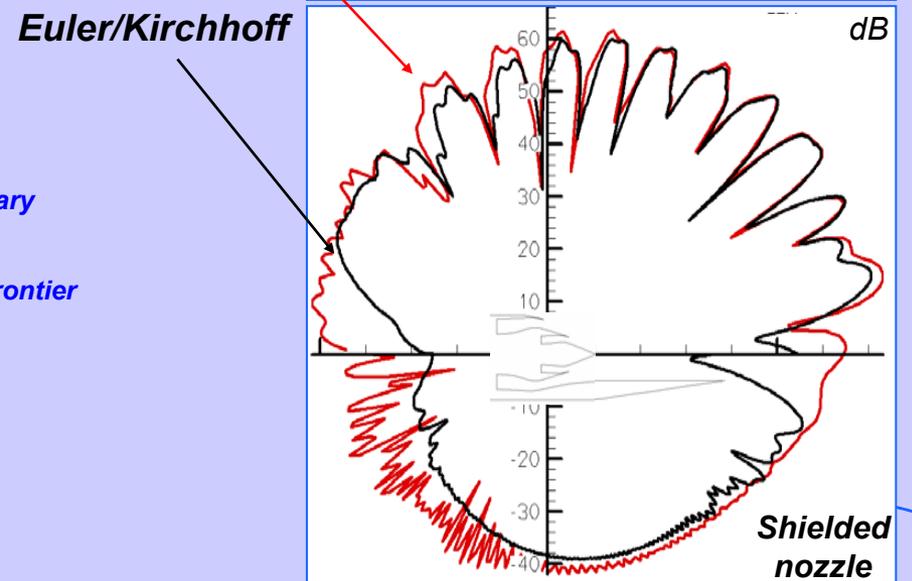
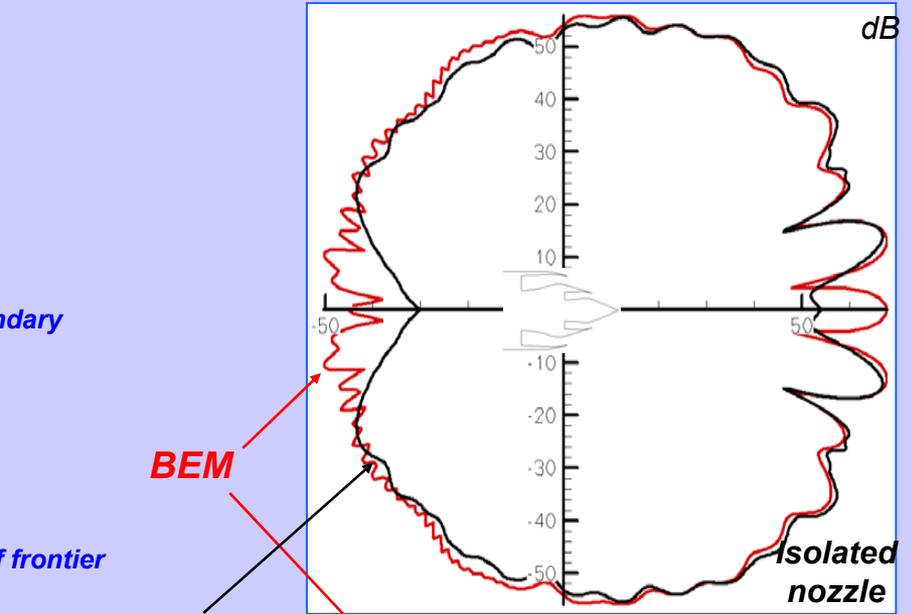
Fan Noise in a Quiescent Medium

(2/3) Far-Field 2D Radiation (Kirch2D code) and Validation (Sysnoise code)

Kirch2D validation



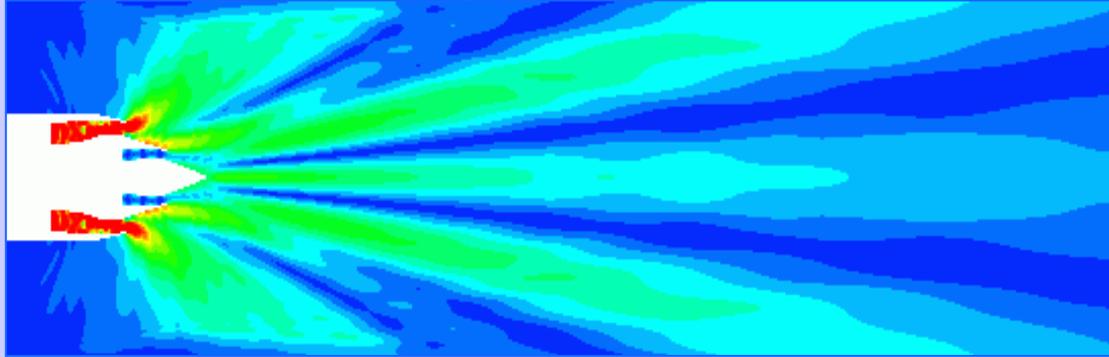
sAbrinA/Kirch2D validation



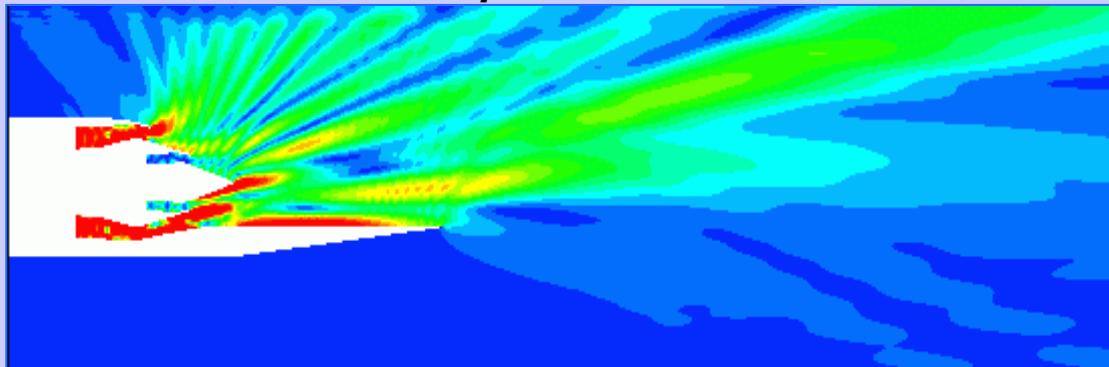
Fan Noise in a Quiescent Medium

(3/3) 2D Shielding Effect Analysis

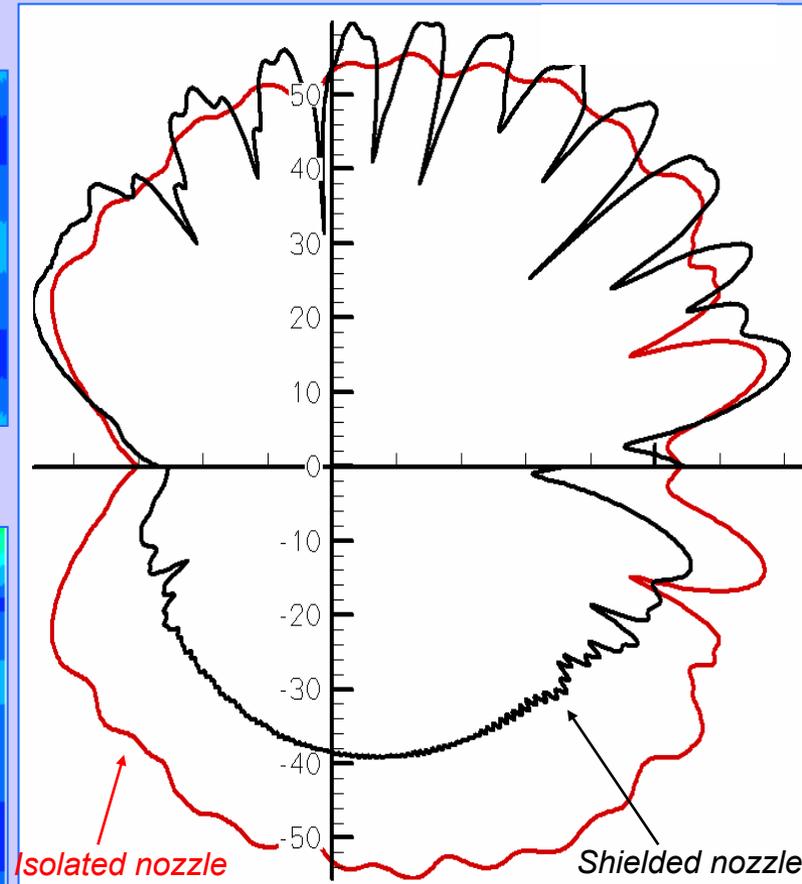
Isolated nozzle : RMS pressure near & mid-fields



Shielded nozzle : RMS pressure near & mid-fields



Far field directivity diagrams (dB)



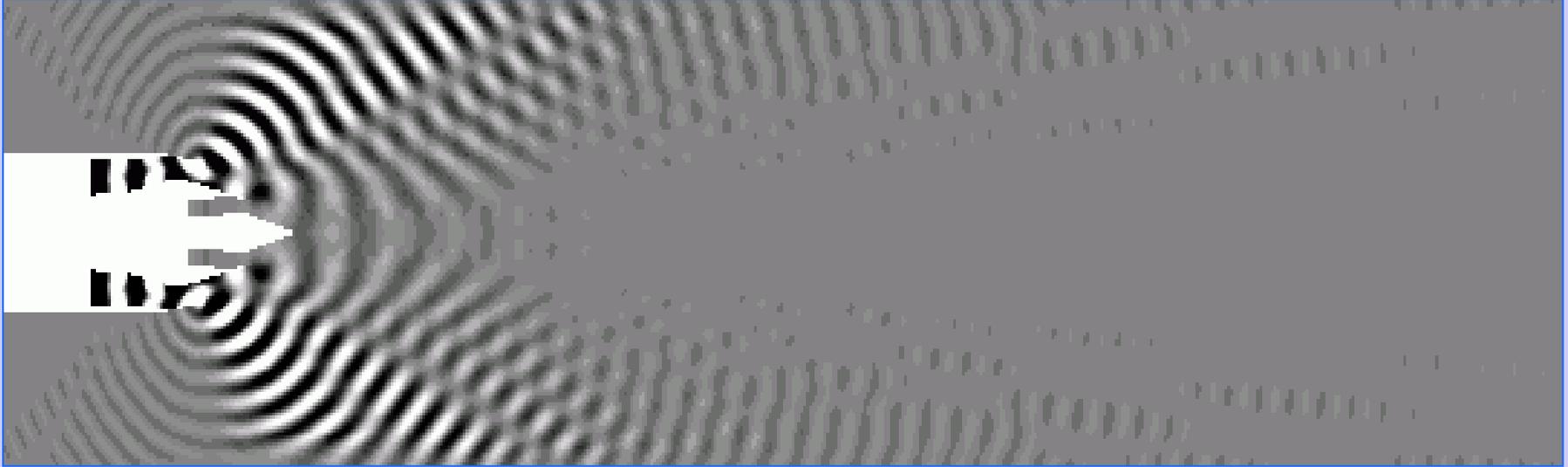
Shielding Effect on the Fan Noise 2D Propagation in an Inhomogeneous Medium

Fan Noise in an Inhomogeneous Medium

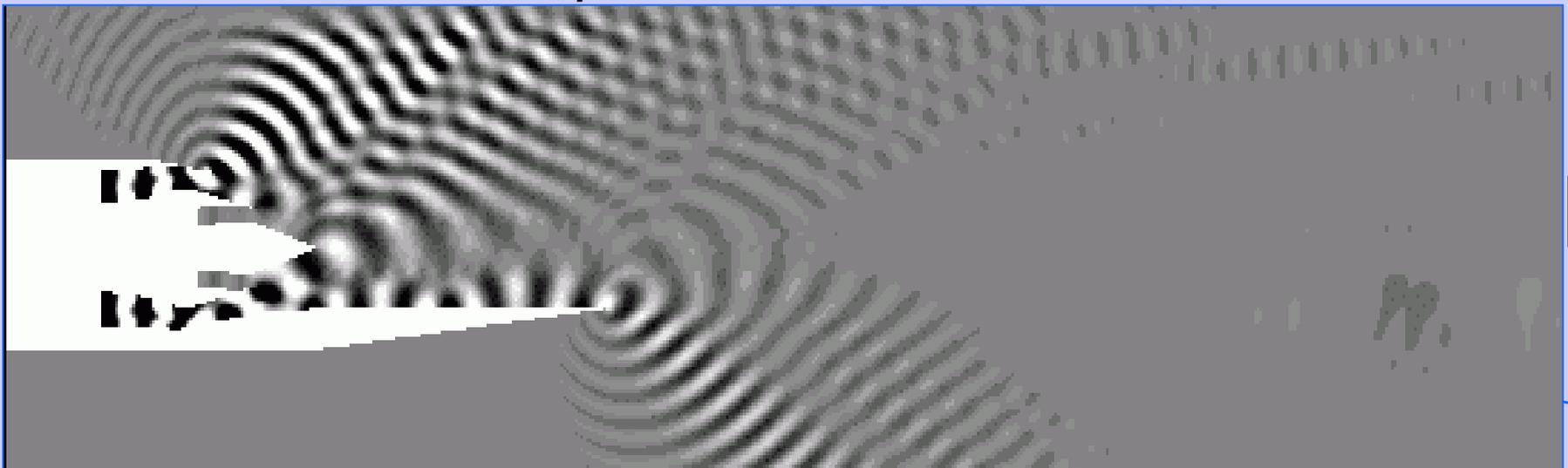
(1/3) Near & Mid-Field 2D Propagation (sAbrinA code)

Isolated nozzle : Instantaneous pressure field

$kR = 20$



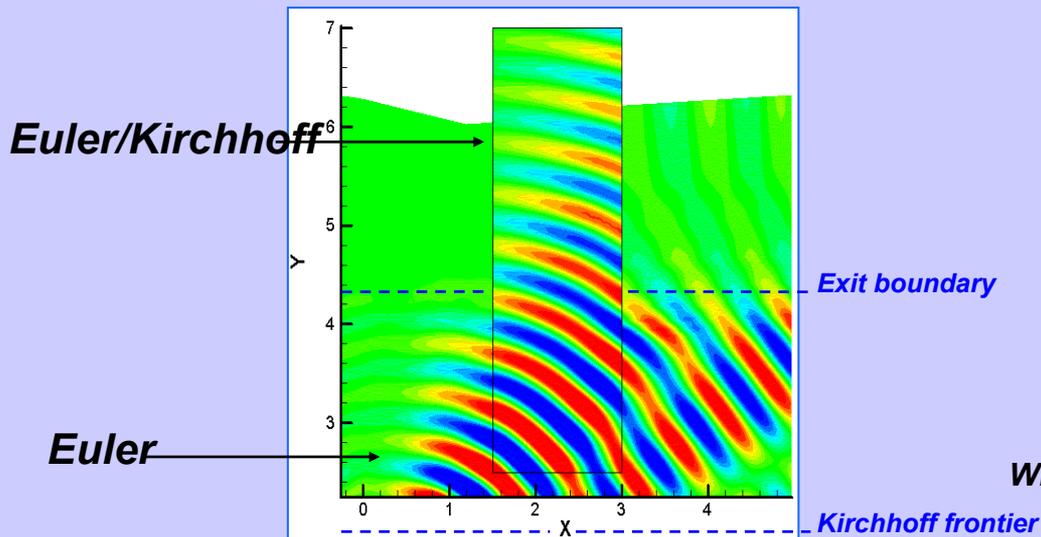
Shielded nozzle: Instantaneous pressure field



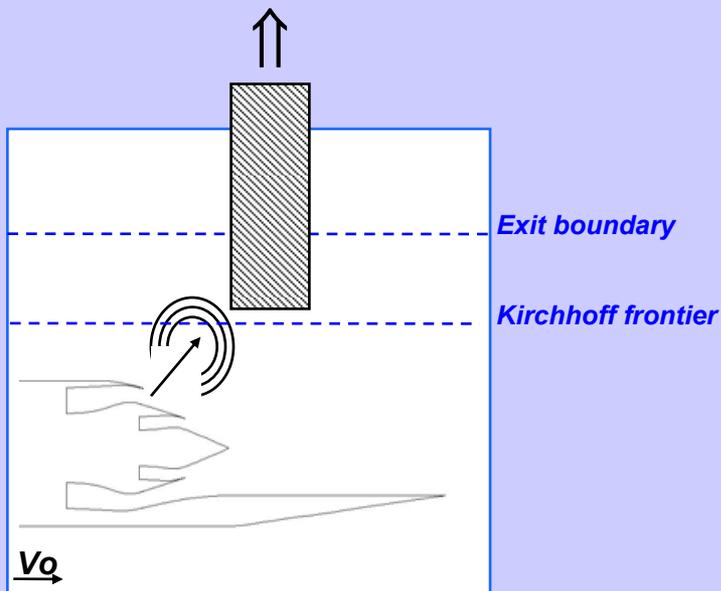
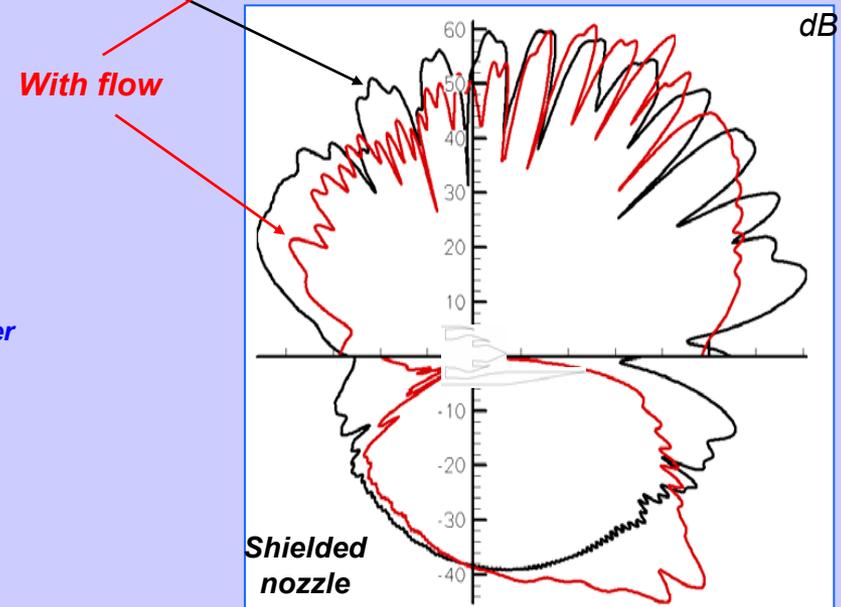
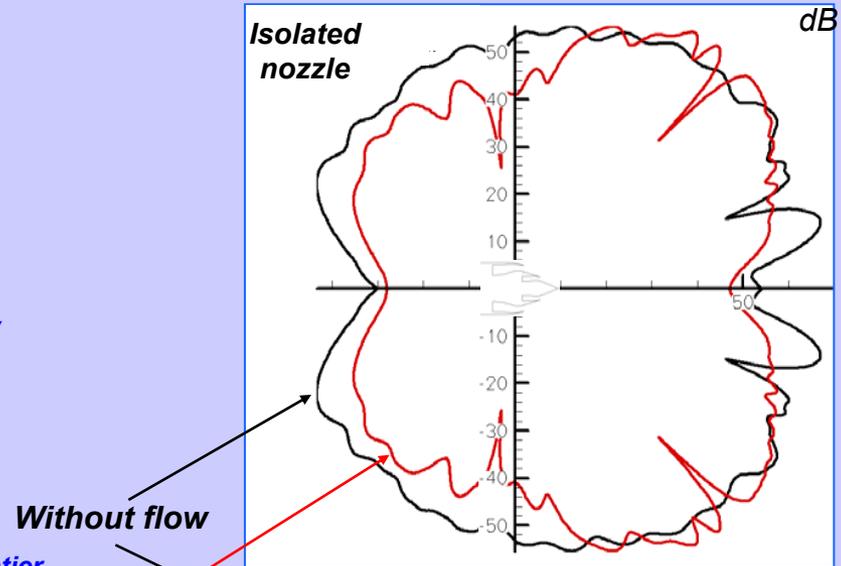
Fan Noise in an Inhomogeneous Medium

(2/3) Far-Field 2D Radiation (Kirch2D code)

Kirch2D validation



sAbrinA/ Kirch2D validation



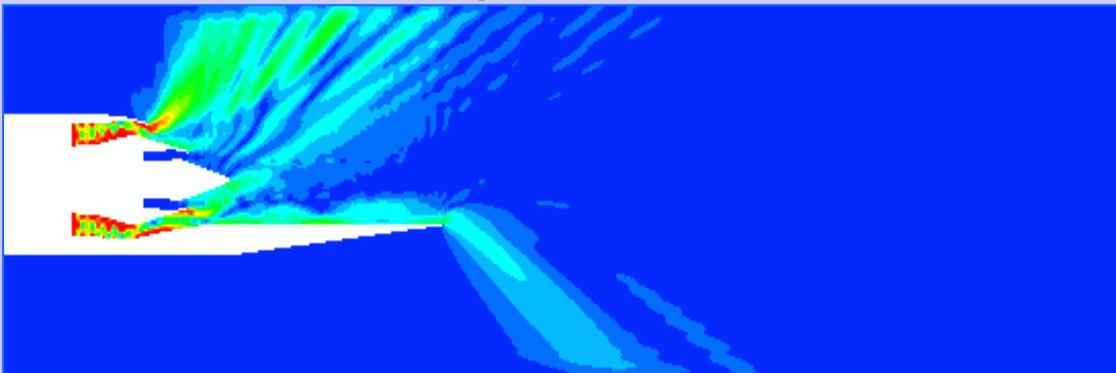
Fan Noise in an Inhomogeneous Medium

(3/3) 2D Shielding Effect Analysis

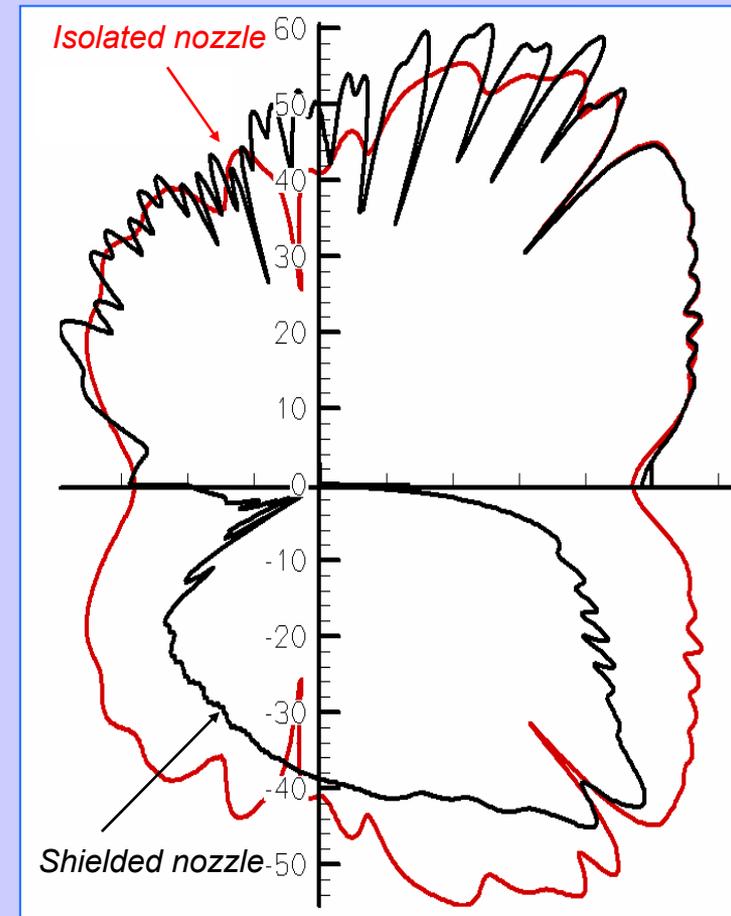
Isolated nozzle : RMS pressure near & mid-fields



Shielded nozzle : RMS pressure near & mid-fields



Far field directivity diagrams (dB)

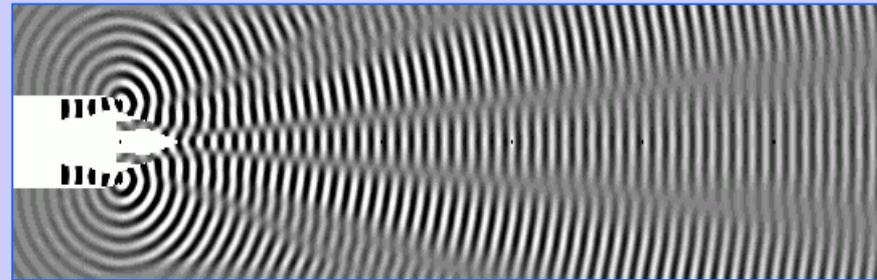


Fan Noise : Conclusions

Shielding & Highly Sheared Mean Flow 2D Effects Analysis

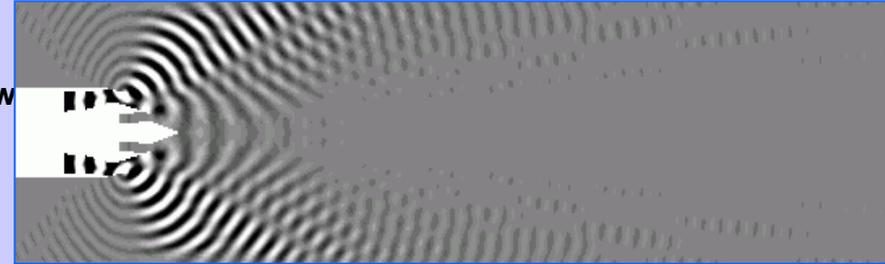
$kR = 20$

Isolated nozzle, without flow



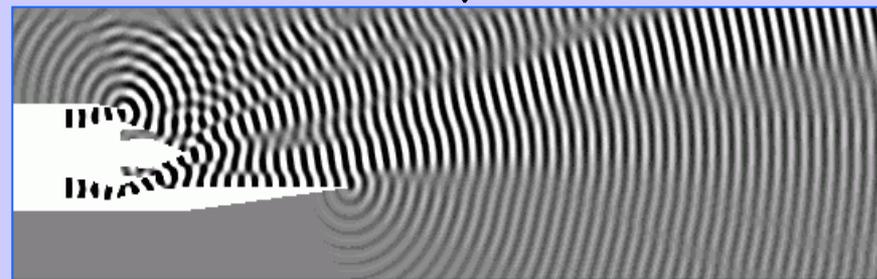
Shielding effect

Isolated nozzle, with flow

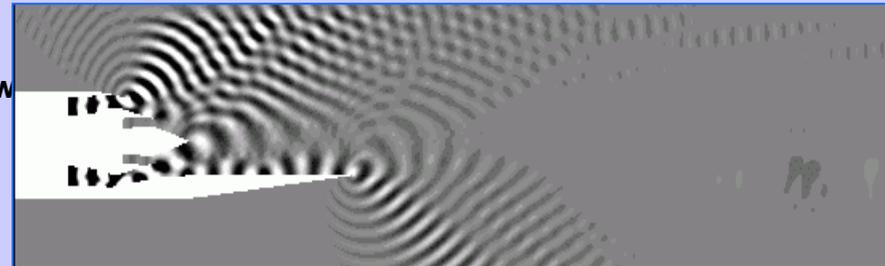


Mean flow effect

Shielding effect



Mean flow effect



Shielded nozzle, with flow

- **Important shielding effect on the rear fan noise in a quiescent medium**
- **Important mean flow refraction effect on the rear fan noise**
- **Significant attenuation of the shielding effect by the mean flow**

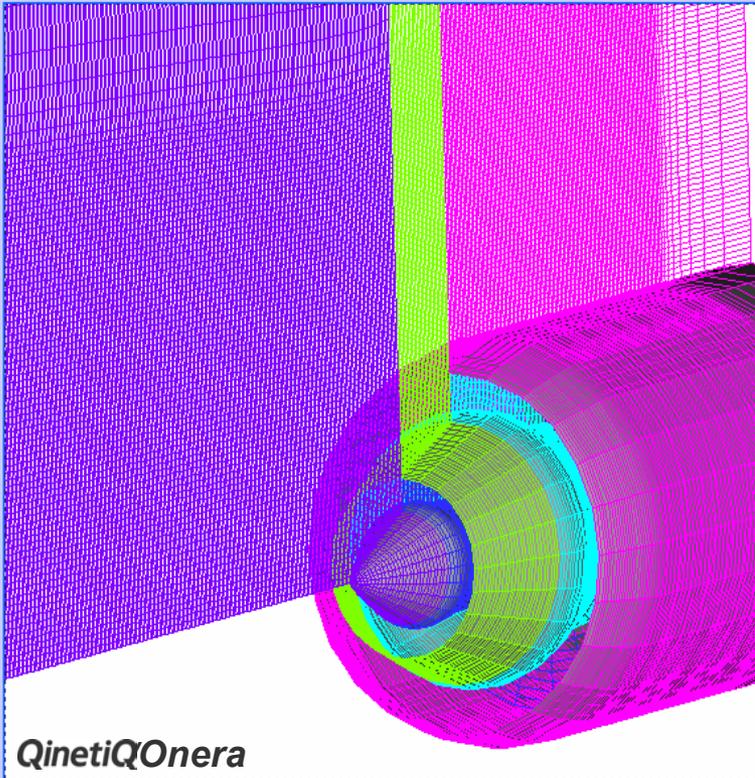
**Fan Noise 3D Propagation in both a
Quiescent and an Inhomogenous Medium,
for the Isolated Nozzle**

Fan Noise for the Isolated Nozzle

Preliminary task : mesh and mean-flow 2D \Rightarrow 3D extension

3D CAA mesh

(1,606,000 points / 25 angular planes)



QinetiQOnera

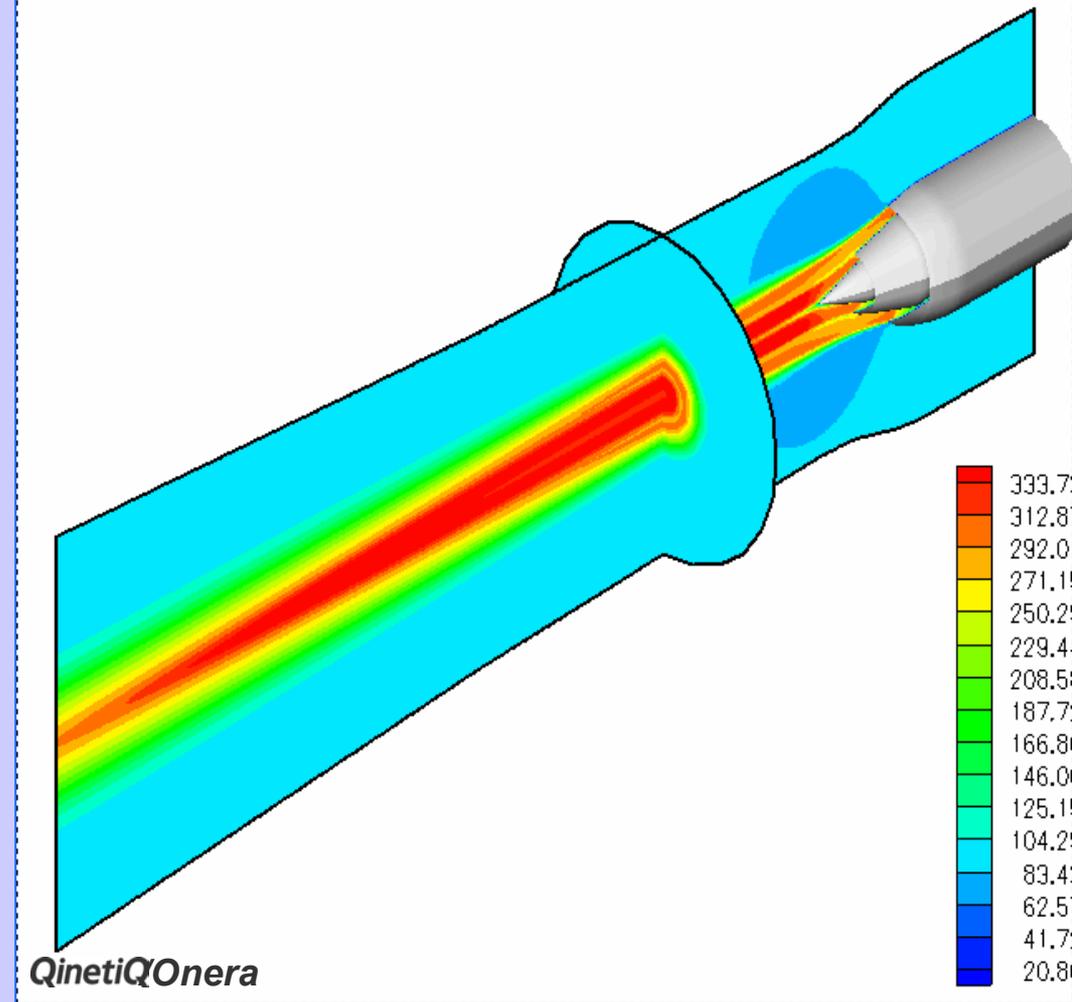
Requirements :

12 points per apparent angular wavelength



spinning modes of order up to 2

3D axi-symmetric mean flow (axial velocity)



QinetiQOnera

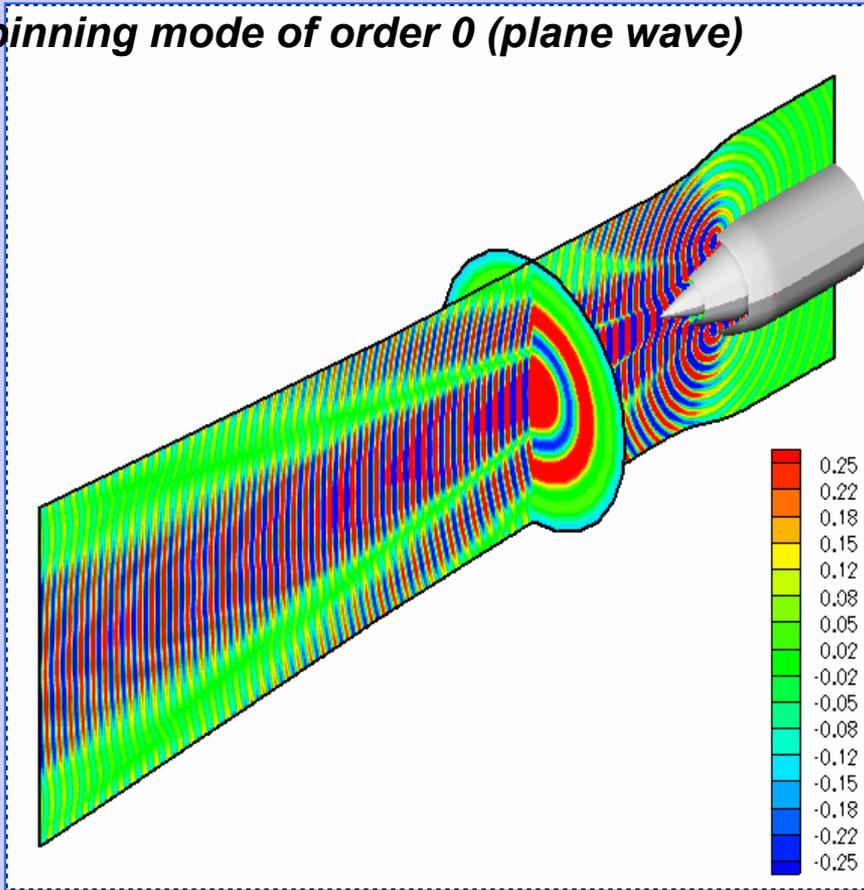
ONERA

Fan Noise in a Quiescent Medium (Isolated Nozzle)

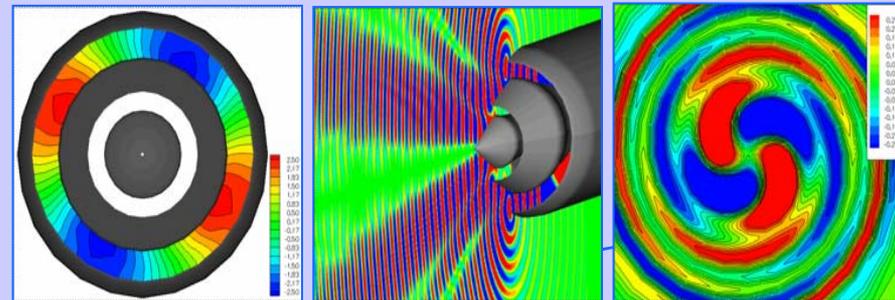
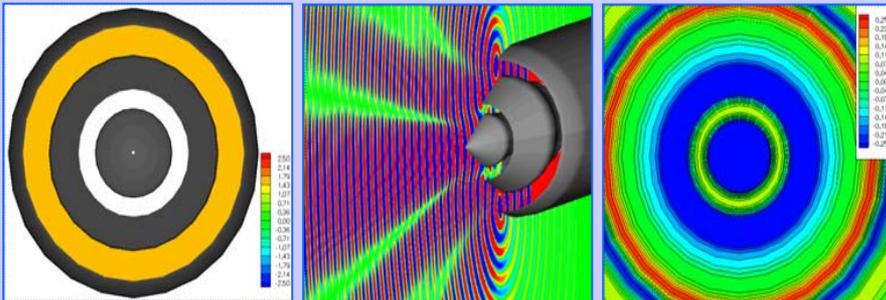
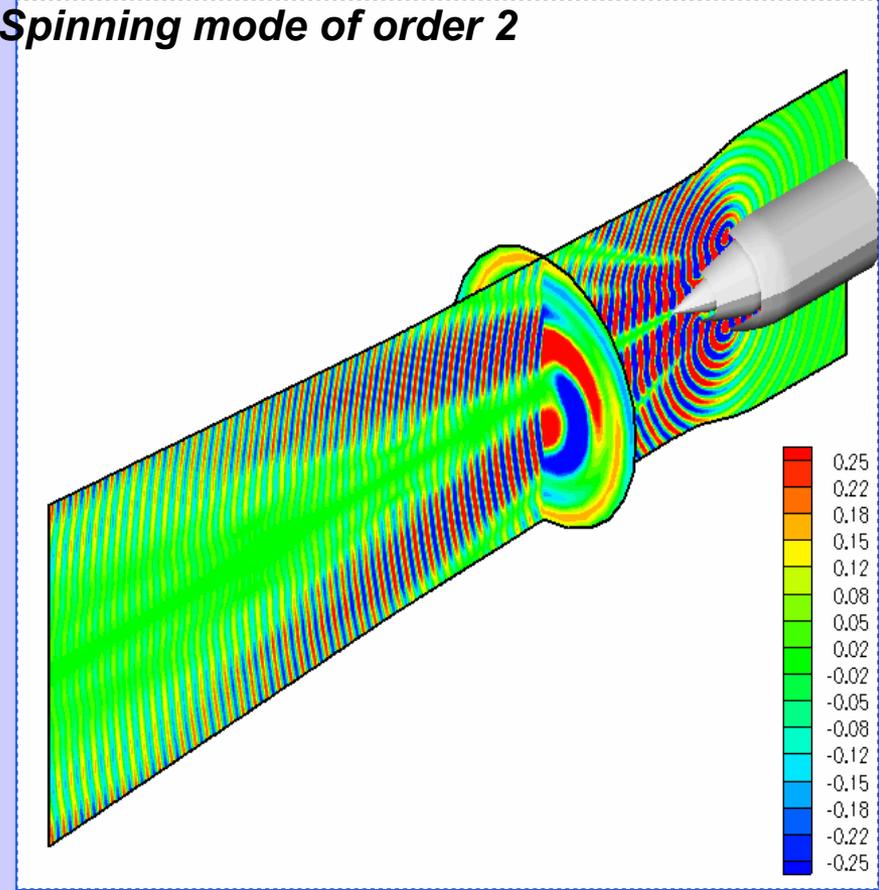
(1/3) Near & Mid-Field 3D Propagation (sAbrinA 3D)



Spinning mode of order 0 (plane wave)



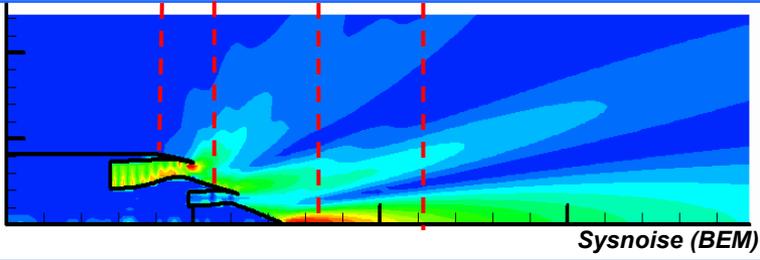
Spinning mode of order 2



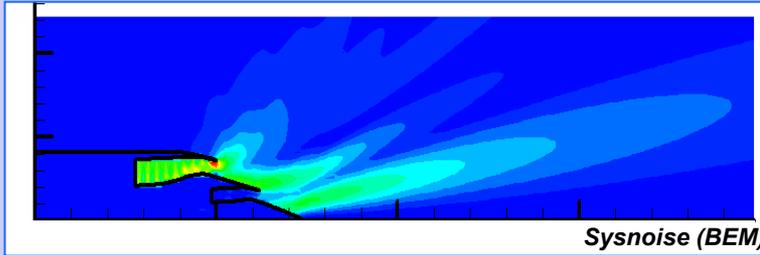
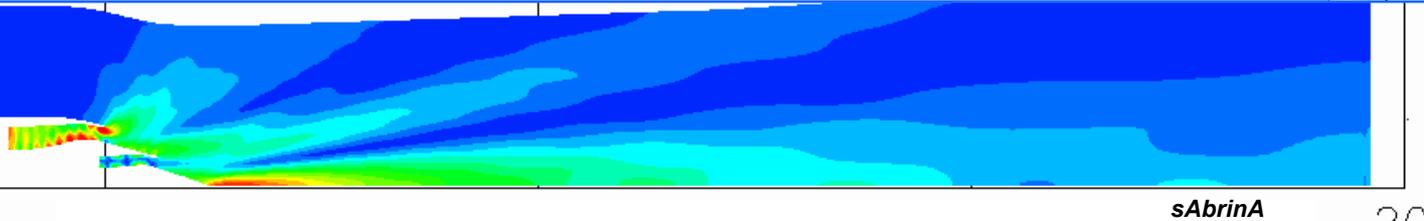
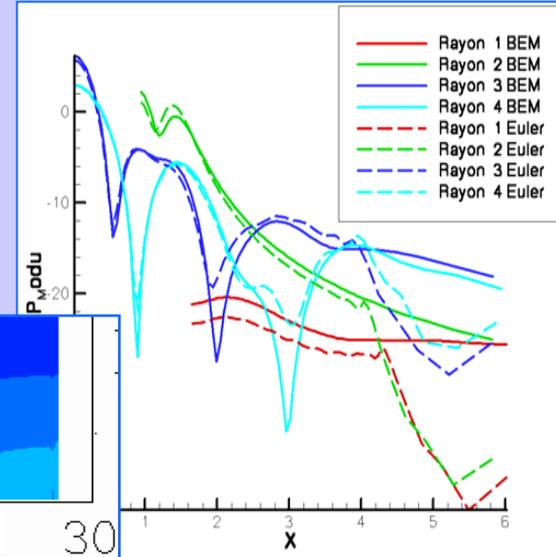
Fan Noise in a Quiescent Medium (Isolated Nozzle)

(2/3) Near & Mid-Field Validation (sAbrinA 3D vs. Sysnoise)

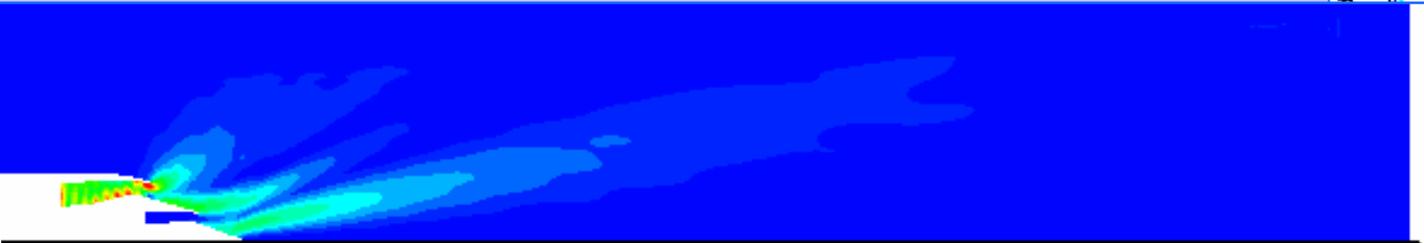
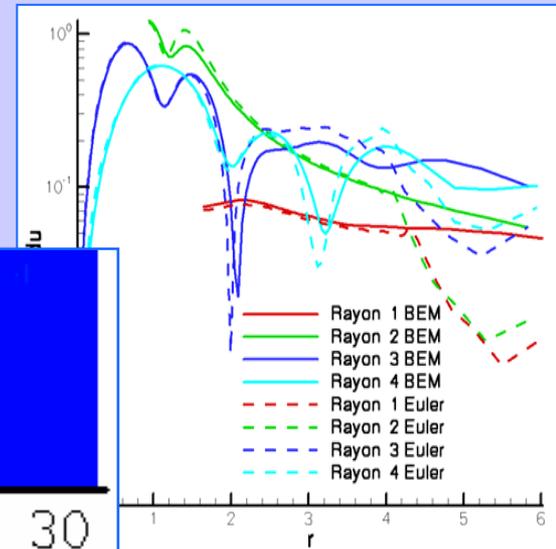
r1 r2 r3 r4



*Spinning mode of order 0
(plane wave)*

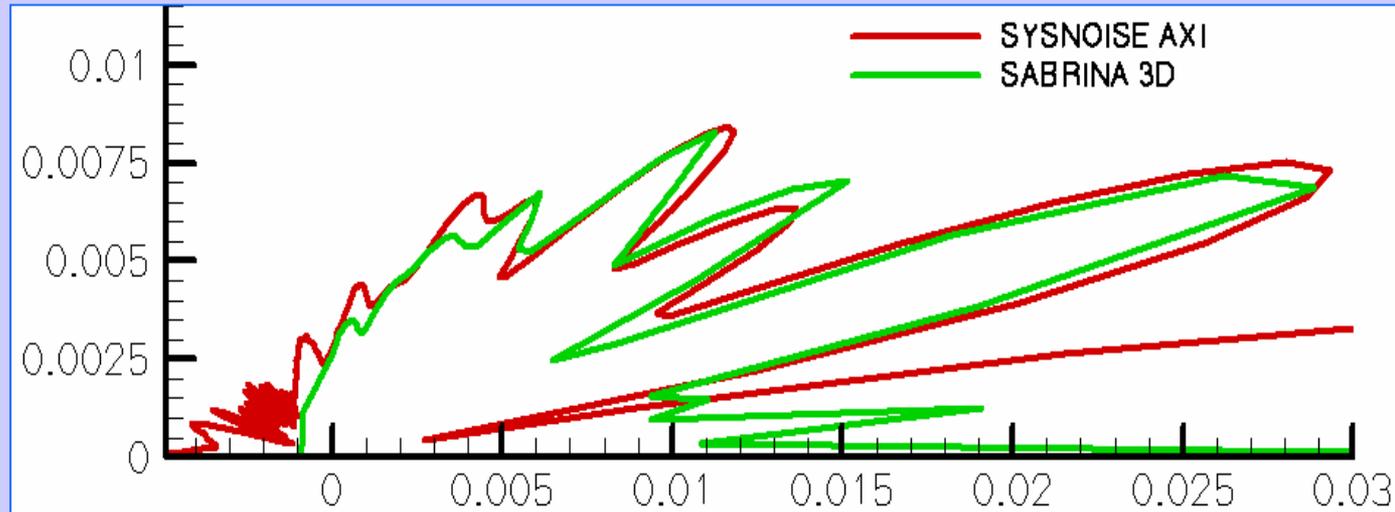


Spinning mode of order 2

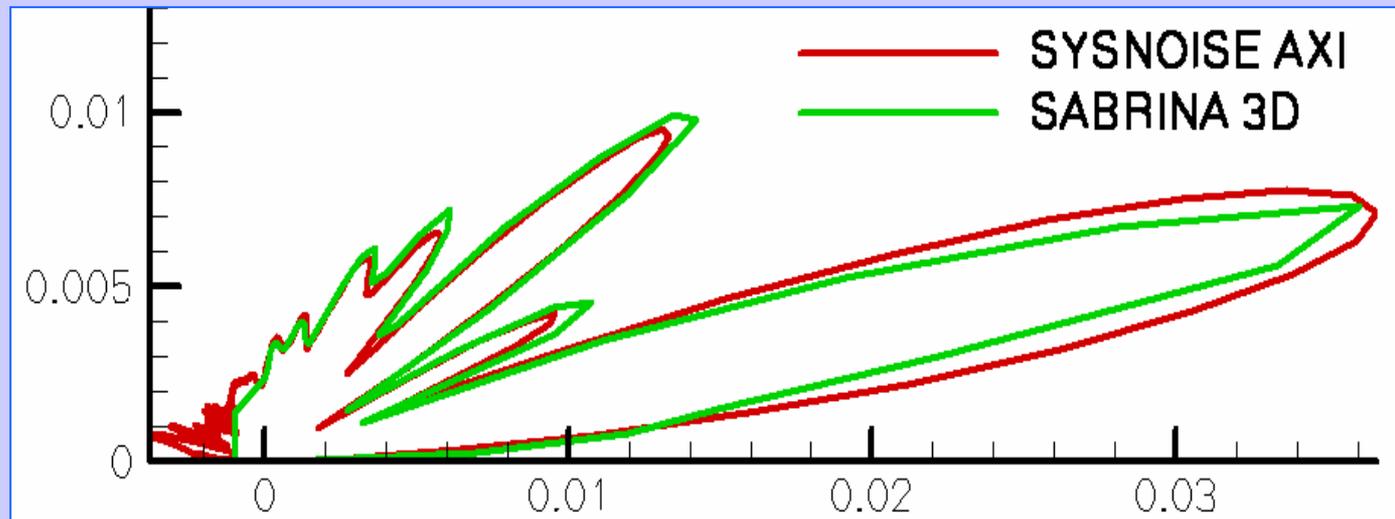


Fan Noise in a Quiescent Medium (Isolated Nozzle)

3/3) Far-Field 3D Radiation & Validation (sAbrinA 3D / Kirch3D vs. Sysnoise)



Spinning mode of order 0 (plane wave)



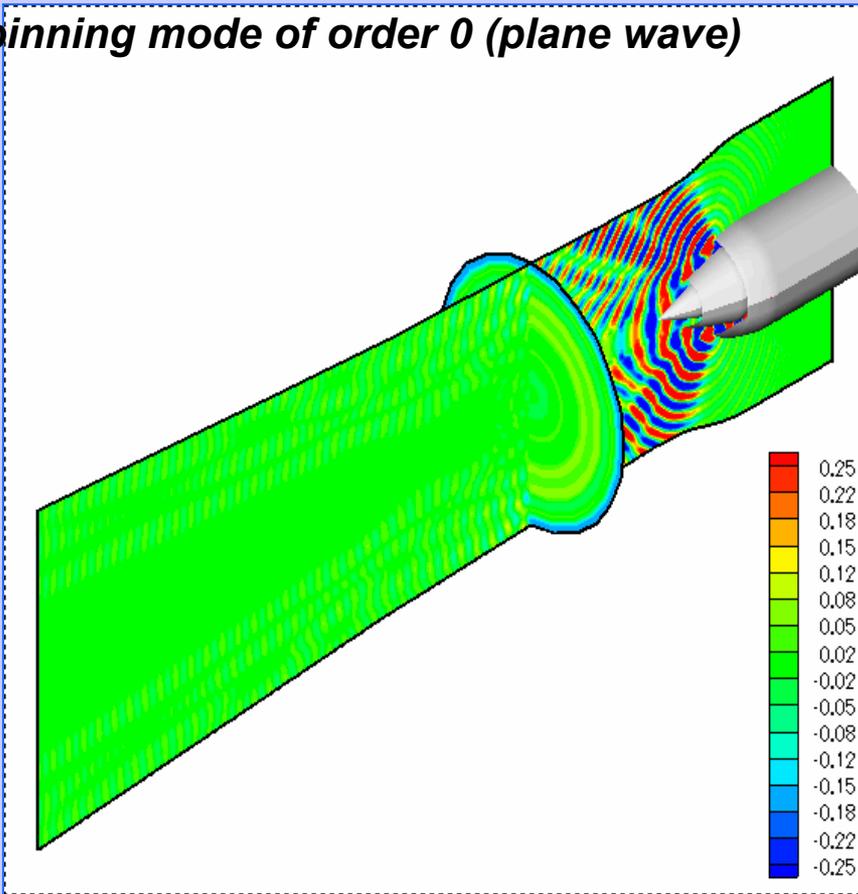
Spinning mode of order 2

Fan Noise in an Inhomogeneous Medium (Isolated Nozzle)

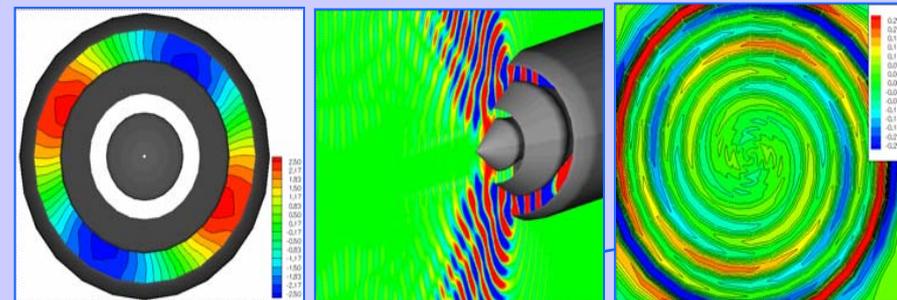
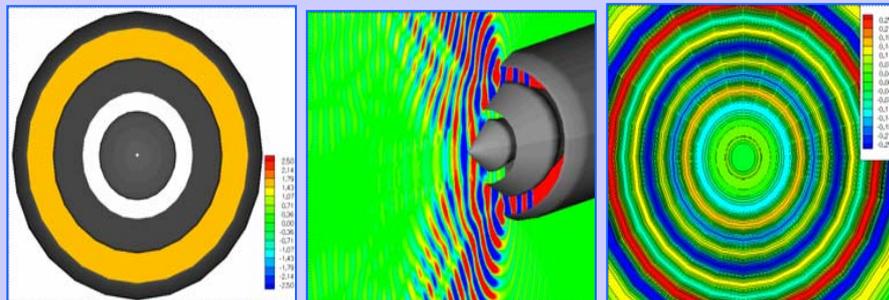
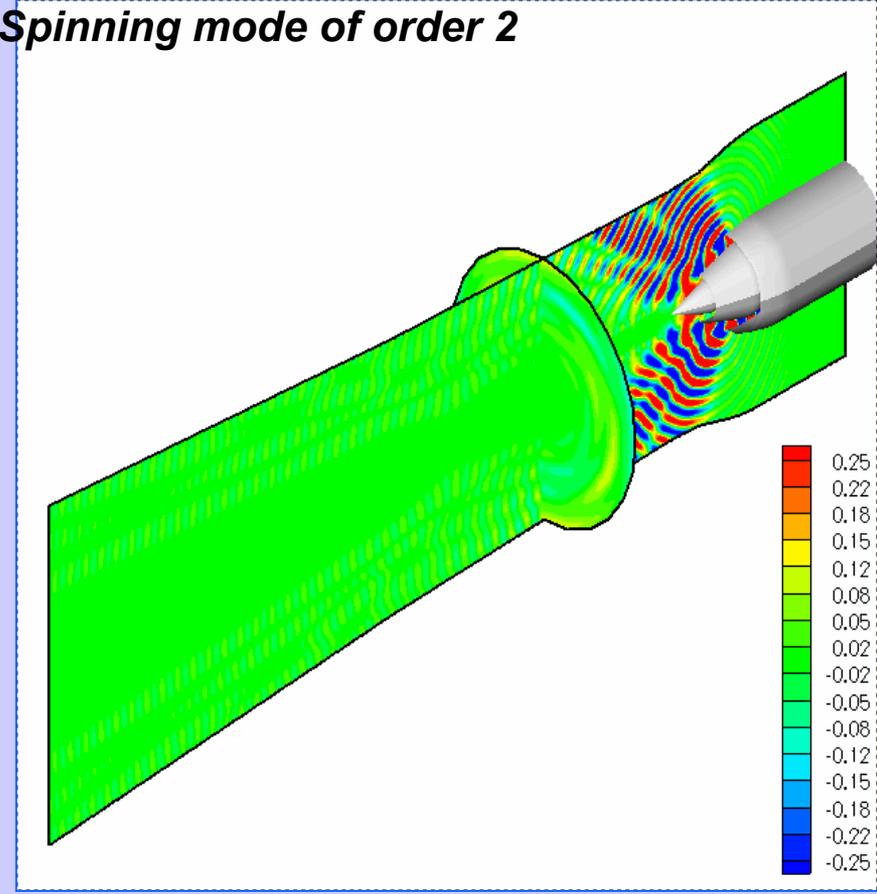
(1/2) Near & Mid-Field 3D Propagation (sAbrinA 3D)



Spinning mode of order 0 (plane wave)

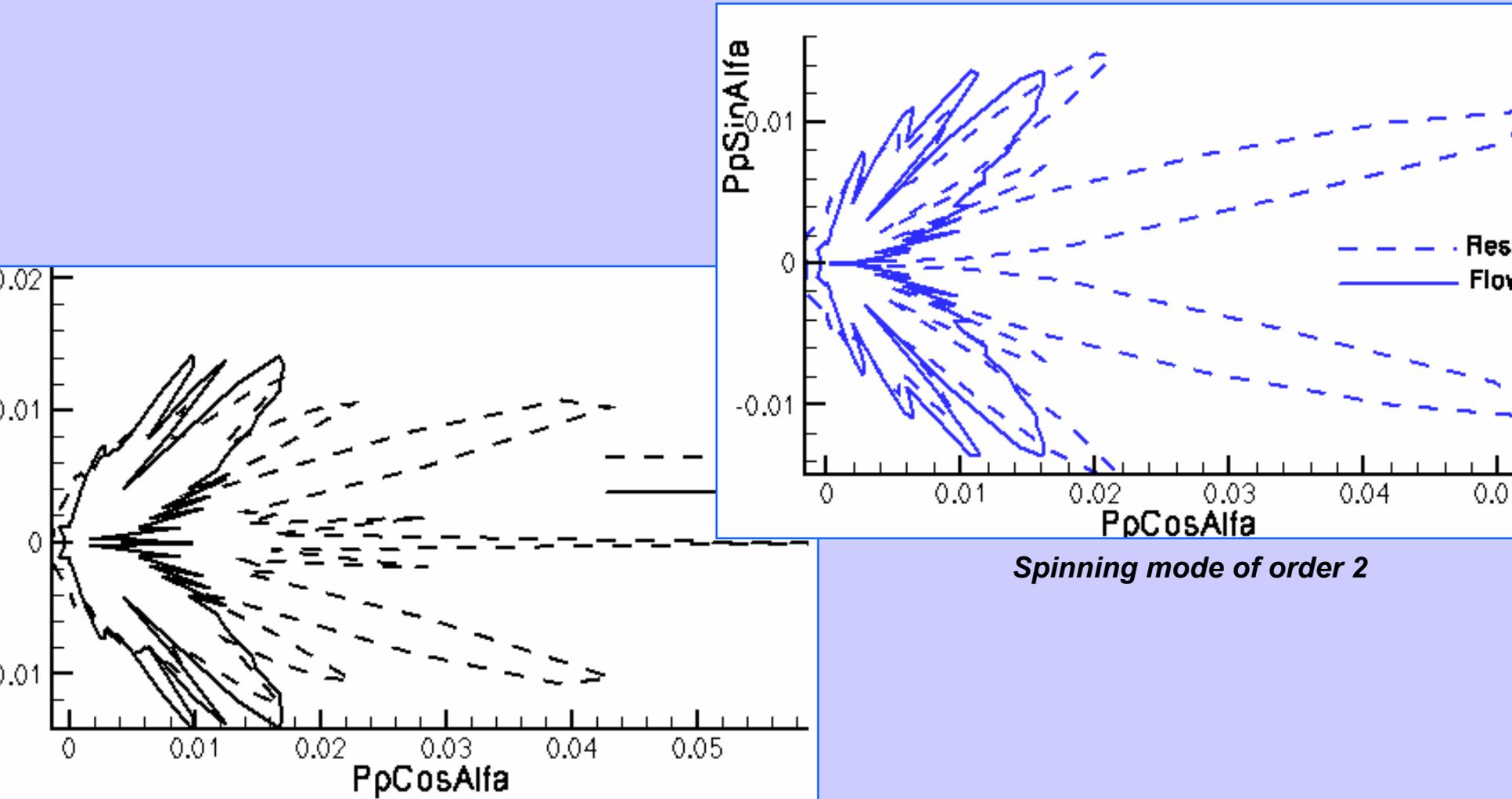


Spinning mode of order 2



Fan Noise in an Inhomogeneous Medium (Isolated Nozzle)

Far-Field 3D Radiation (sAbrinA 3D/ Kirch3D), and Mean Flow Effects Analy



Spinning mode of order 0 (plane wave)

Spinning mode of order 2

Conclusions

- ❑ **Validation of the hybrid methodology & tools used for this kind of engine applications**
- ❑ **Important shielding effect on the 2D propagation in a medium at rest**
- ❑ **Important (highly sheared) mean flow refraction effect on the 2D propagation**
- ❑ **Significant attenuation of the 2D shielding effect by the (inhomogeneous) mean flow**
- ❑ **Extension of the isolated nozzle study to 3D : still important mean flow refraction effects**

Perspectives

- ❑ **3D computations over the shielded nozzle (complex 3D geometry)**
- ❑ **Generation of exact fan noise modes, in order to perform a more realistic calculation (European Project “NACRE”)**